

22. Wellen

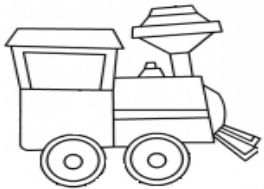
Überlagerung von Wellen

- **Reflektion und Transmission von Wellen an Grenzflächen**
- **Ausprägungen und Eigenschaften stehender Wellen**
- **Überlagerung der Bewegung von Beobachter und Wellenquelle**

Dopplereffekt

1. Wellenberg verlässt Zug

t_0



$\vec{r}_1(t_0)$

$\vec{r}_2(t_1)$



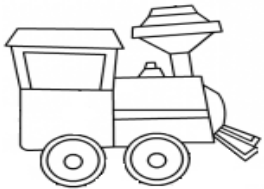
t_1 1. Wellenberg kommt an

$$|\vec{r}_1(t_0) - \vec{r}_2(t_1)| = c(t_1 - t_0)$$

Dopplereffekt

1. Wellenberg verlässt Zug

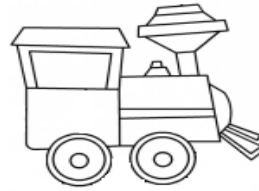
t_0



$\vec{r}_1(t_0)$

2. Wellenberg verlässt Zug

$t_0 + T$



$\vec{r}_1(t_0 + T)$

$$f_{\text{Quelle}} = \frac{1}{T}$$

t_2 2. Wellenberg kommt an

$\vec{r}_2(t_2)$

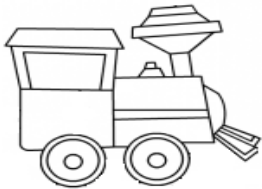


$$|\vec{r}_1(t_0 + T) - \vec{r}_2(t_2)| = c(t_2 - t_0 - T)$$

Dopplereffekt

1. Wellenberg verlässt Zug

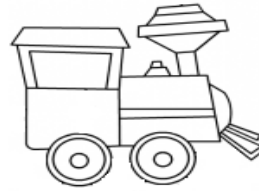
t_0



$\vec{r}_1(t_0)$

2. Wellenberg verlässt Zug

$t_0 + T$



$\vec{r}_1(t_0 + T)$

$$f_{\text{Quelle}} = \frac{1}{T}$$

$$\tilde{f}_{\text{Beob}} = \frac{1}{t_2 - t_1}$$

$\vec{r}_2(t_1)$



t_1 1. Wellenberg kommt an

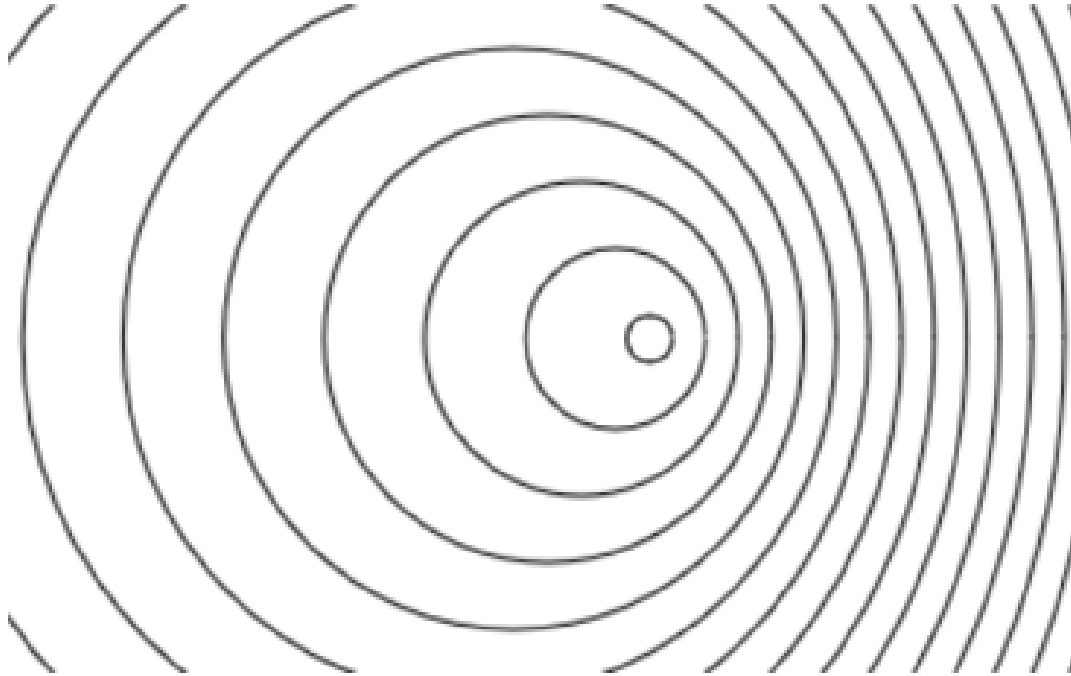
t_2 2. Wellenberg kommt an

$\vec{r}_2(t_2)$



Beobachter nimmt eine modifizierte Frequenz wahr

Dopplereffekt



$$f_{\text{Quelle}} = \frac{1}{T}$$

$$\tilde{f}_{\text{Beob}} = \frac{1}{t_2 - t_1}$$

modifizierte Frequenz durch Eintreffzeiten benachbarter Wellenberge bestimmt

$$|\vec{r}_1(t_0) - \vec{r}_2(t_1)| = c(t_1 - t_0)$$

$$|\vec{r}_1(t_0 + T) - \vec{r}_2(t_2)| = c(t_2 - t_0 - T)$$

Dopplereffekt - Spezialfälle

Quelle Beobachter beobachtete Frequenz

• ← • $f_B = f_Q \left(1 + \frac{v_B}{c}\right)$ (5.205)

• • → $f_B = f_Q \left(1 - \frac{v_B}{c}\right)$ (5.206)

• → • $f_B = \frac{f_Q}{1 - \frac{v_Q}{c}}$ (5.207)

← • • $f_B = \frac{f_Q}{1 + \frac{v_Q}{c}}$ (5.208)

• → ← • $f_B = f_Q \frac{c + v_B}{c - v_Q}$ (5.209)

← • • → $f_B = f_Q \frac{c - v_B}{c + v_Q}$ (5.210)

← • ← • $f_B = f_Q \frac{c + v_B}{c + v_Q}$ (5.211)

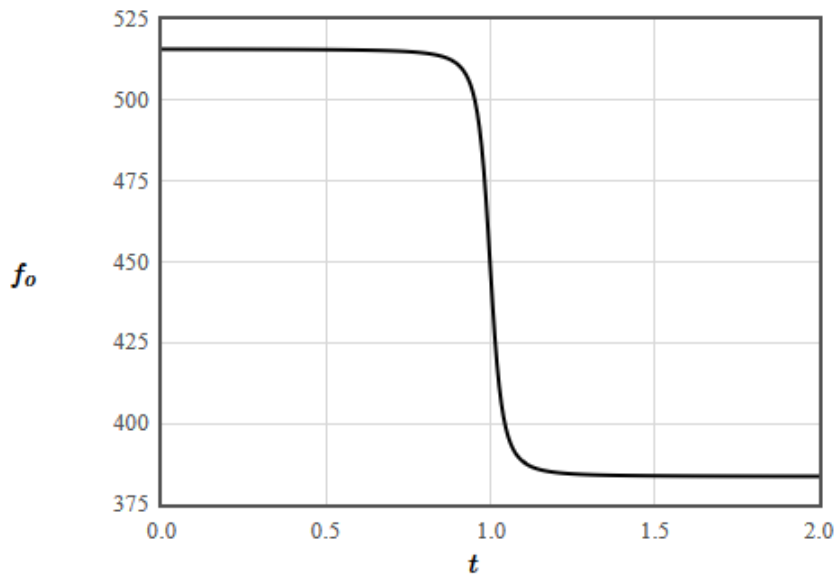
• → • → $f_B = f_Q \frac{c - v_B}{c - v_Q}$ (5.212)

Dopplereffekt

$$|\vec{r}_o(t_1) - \vec{r}_s(t_0)| = c(t_1 - t_0),$$

$$|\vec{r}_o(t_2) - \vec{r}_s(t_0 + T)| = c(t_2 - t_0 - T),$$

$$f_o = \frac{1}{t_2 - t_1}.$$



$$\vec{r}_s(t) = 50*t-50 \hat{x} + 0 \hat{y} + 0 \hat{z} \text{ [m].}$$

$$\vec{r}_o(t) = 0 \hat{x} + 2 \hat{y} + 0 \hat{z} \text{ [m].}$$

$$f_s = 440 \text{ [Hz]} \quad c = 340 \text{ [m/s]}$$

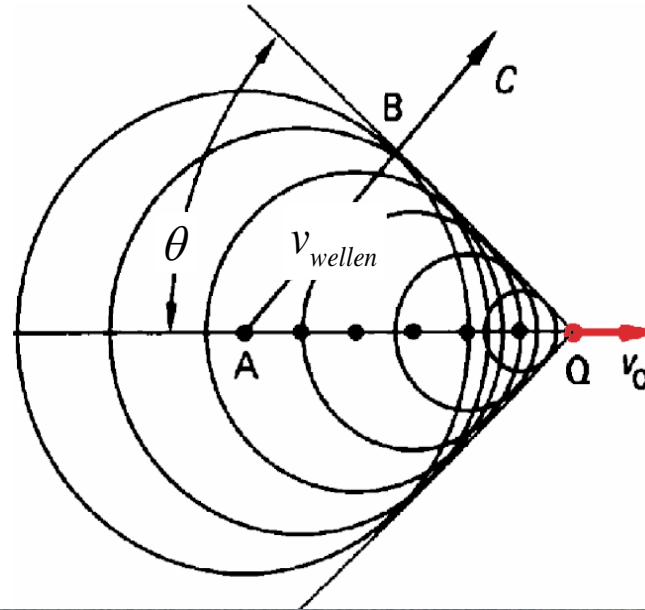
Plot f_o from $t = 0$ to $t = 2$.

At time $t = 0$ s, the observer hears a frequency of 515.8 Hz.

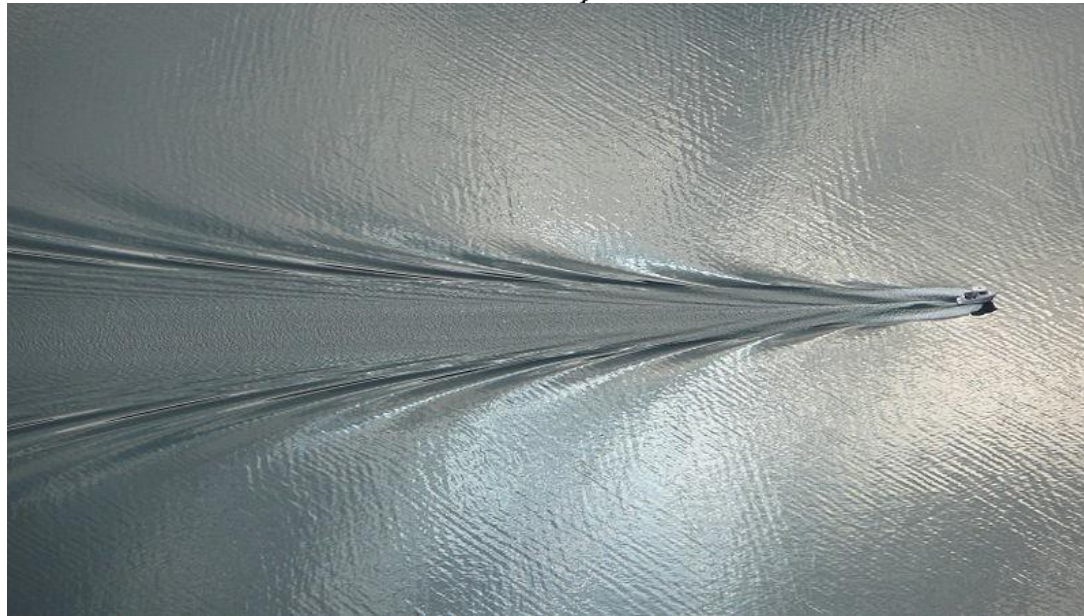
At time $t = 2$ s, the observer hears a frequency of 383.6 Hz.

Überschallgeschwindigkeit

$$\sin \theta = \frac{v_{\text{wellen}}}{v_{\text{flugzeug}}}$$

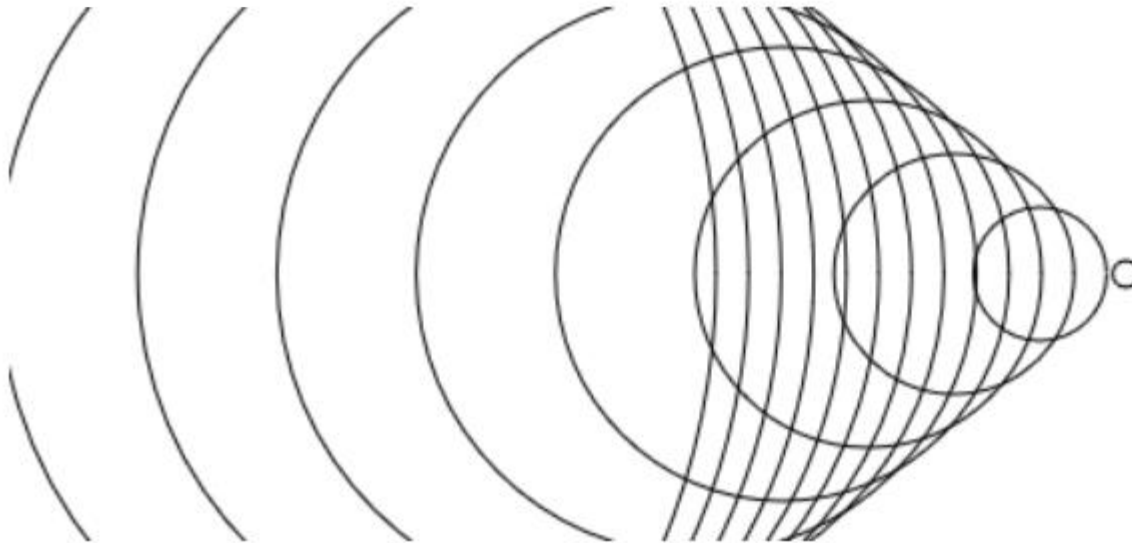


Hering



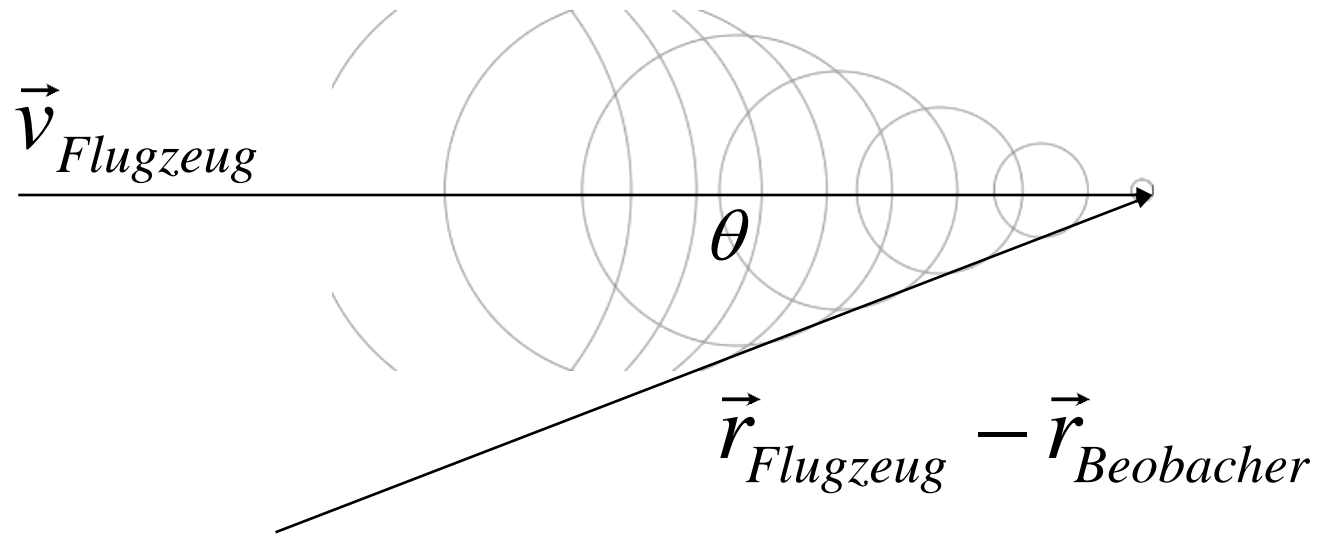
$$\text{Mach-Zahl} = \frac{v}{v_{\text{wellen}}}$$

Bewegte Wellenquelle



Ernst Mach

Überschallgeschwindigkeit



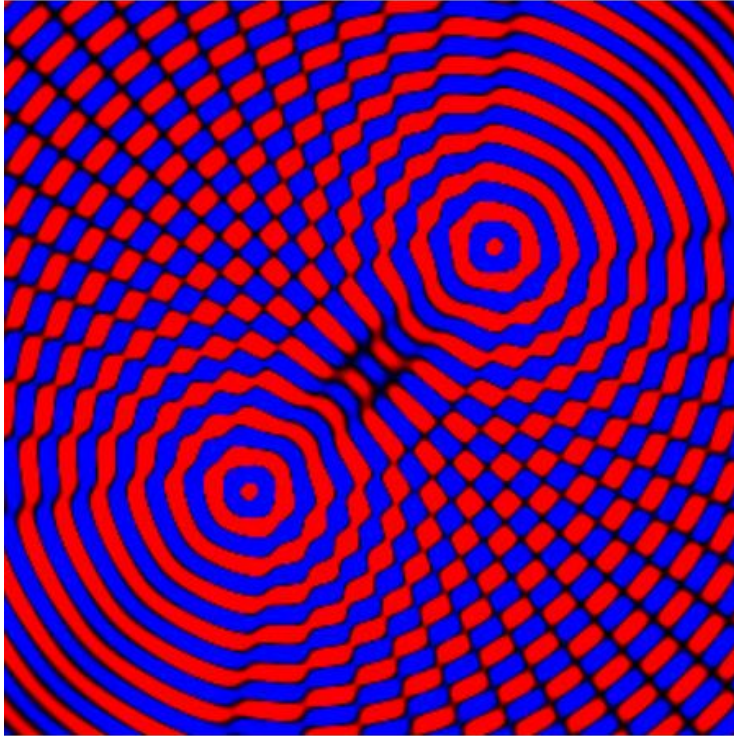
$$\sin \theta = \frac{|\vec{v}_{wellen}|}{|\vec{v}_{flugzeug}|}$$

$$\cos \theta = \frac{\vec{v}_{Flugzeug} \cdot (\vec{r}_{Flugzeug} - \vec{r}_{Beobachter})}{|\vec{v}_{flugzeug}| |\vec{r}_{Flugzeug} - \vec{r}_{Beobachter}|}$$

Überlagerung von Wellen

- ❑ **Überlagerung von mehreren Wellen**
 - ❑ **Interferenz**
 - ❑ **Beugung**

Interferenz zweier Oberflächenwellen



$A_1 =$	<input type="text" value="0.1"/>	[cm ²]	$A_2 =$	<input type="text" value="0.1"/>	[cm ²]
$x_1 =$	<input type="text" value="2"/>	[cm]	$x_2 =$	<input type="text" value="4"/>	[cm]
$y_1 =$	<input type="text" value="2"/>	[cm]	$y_2 =$	<input type="text" value="4"/>	[cm]
$\phi_1 =$	<input type="text" value="0"/>	[rad]	$\phi_2 =$	<input type="text" value="0"/>	[rad]

$\lambda =$	<input type="text" value="0.3"/>	[cm]	$T =$	<input type="text" value="0.5"/>	[s]
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at $t =$ [s].

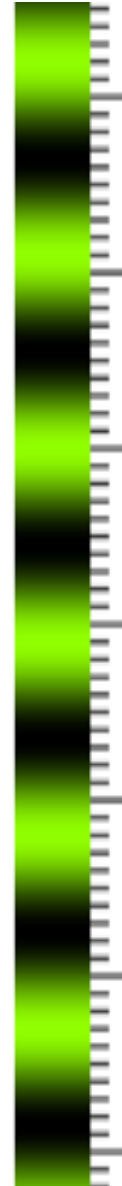
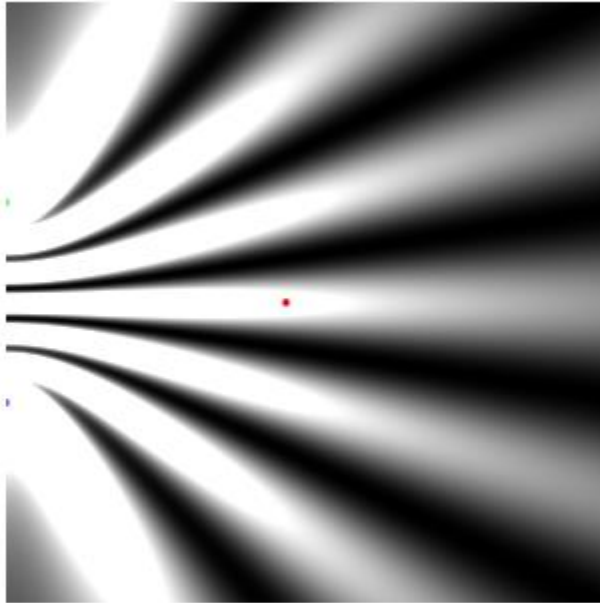
$$c = \frac{\lambda}{T}$$

$$z(r, t) = \frac{A_1}{\sqrt{|r - r_1|}} \cos(k|r - r_1| - \omega t + \phi_1) + \frac{A_2}{\sqrt{|r - r_2|}} \cos(k|r - r_2| - \omega t + \phi_2)$$

Interferenz zweier Oberflächenwellen

$$z(r, t) = \frac{A_1}{\sqrt{|r - r_1|}} \cos(k|r - r_1| - \omega t + \varphi_1) + \frac{A_2}{\sqrt{|r - r_2|}} \cos(k|r - r_2| - \omega t + \varphi_2)$$

Doppelspalt



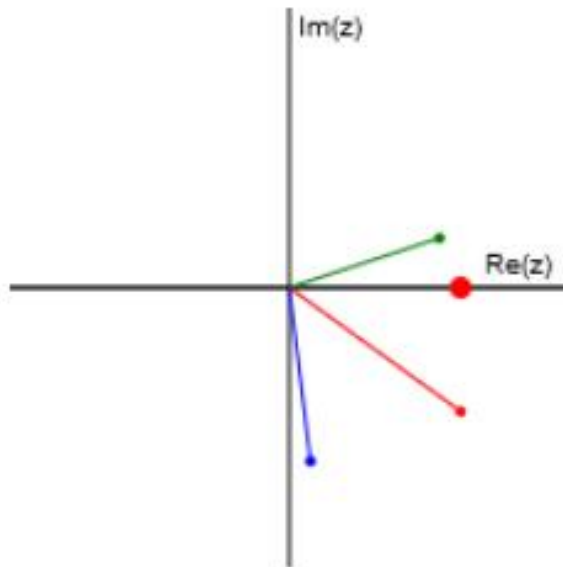
Gesamtamplitude A

$$A = \sum_{j=1}^N \frac{A_j}{r_j} e^{i(k|\vec{r}-\vec{r}_j|+\phi_j)}$$

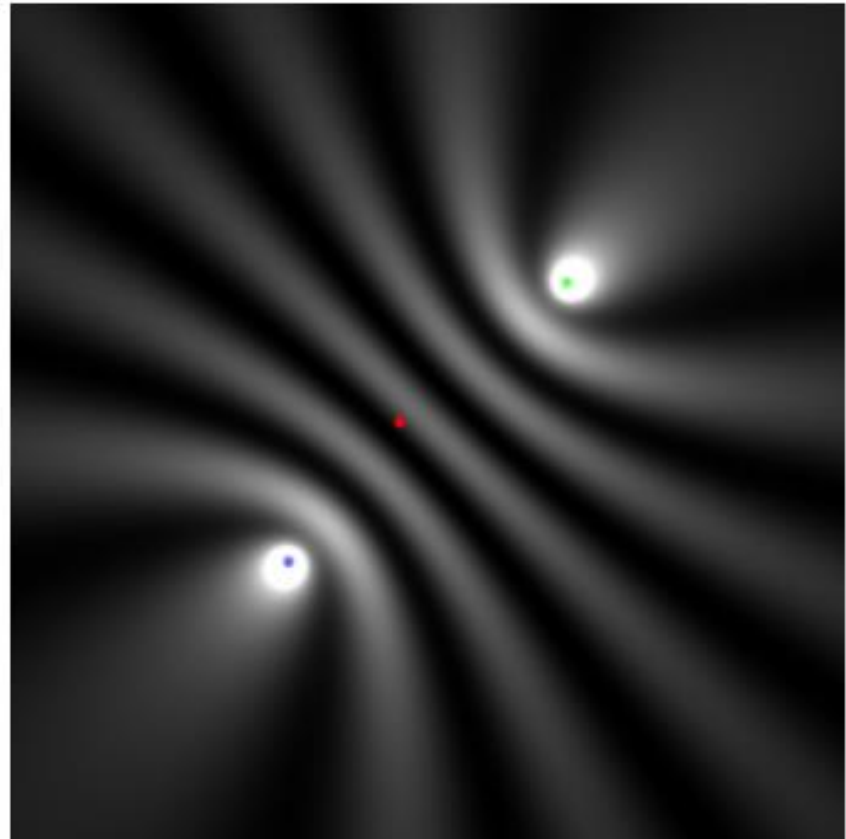
Intensität

$$I \propto A^* A$$

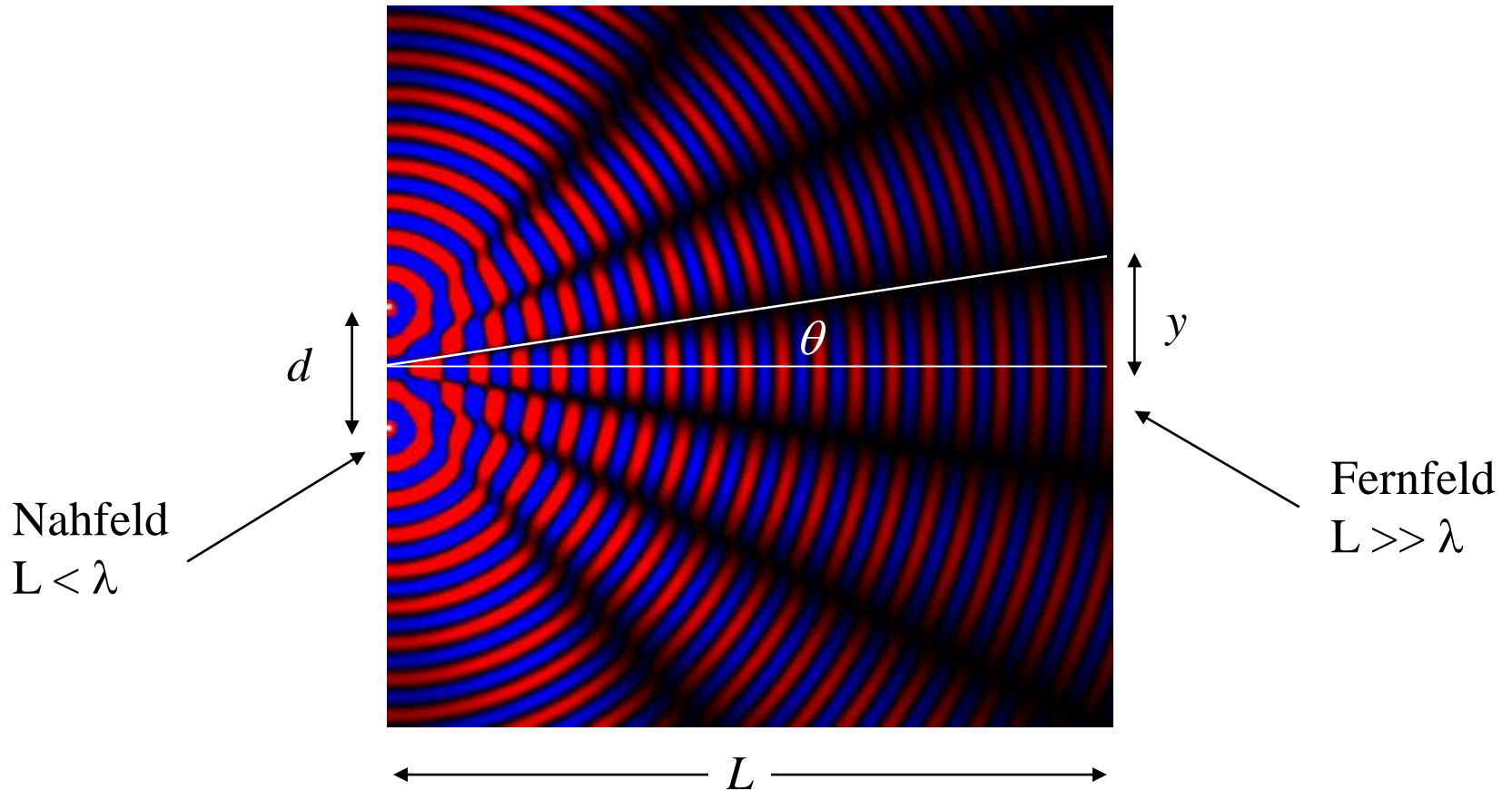
Intensität interferierender Oberflächenwellen



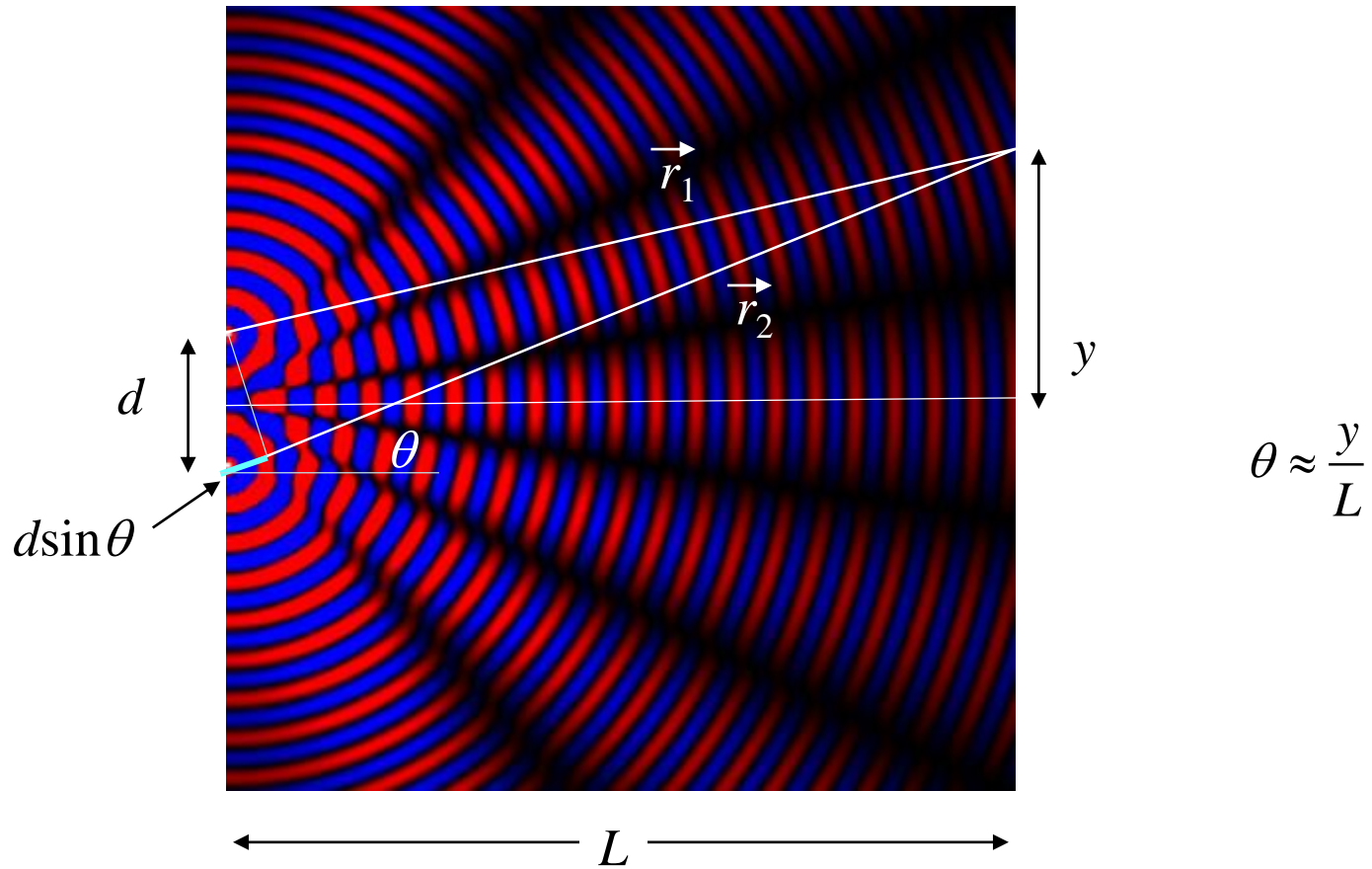
$$|A| = 1.07 \text{ [cm]}$$



Nahfeld / Fernfeld



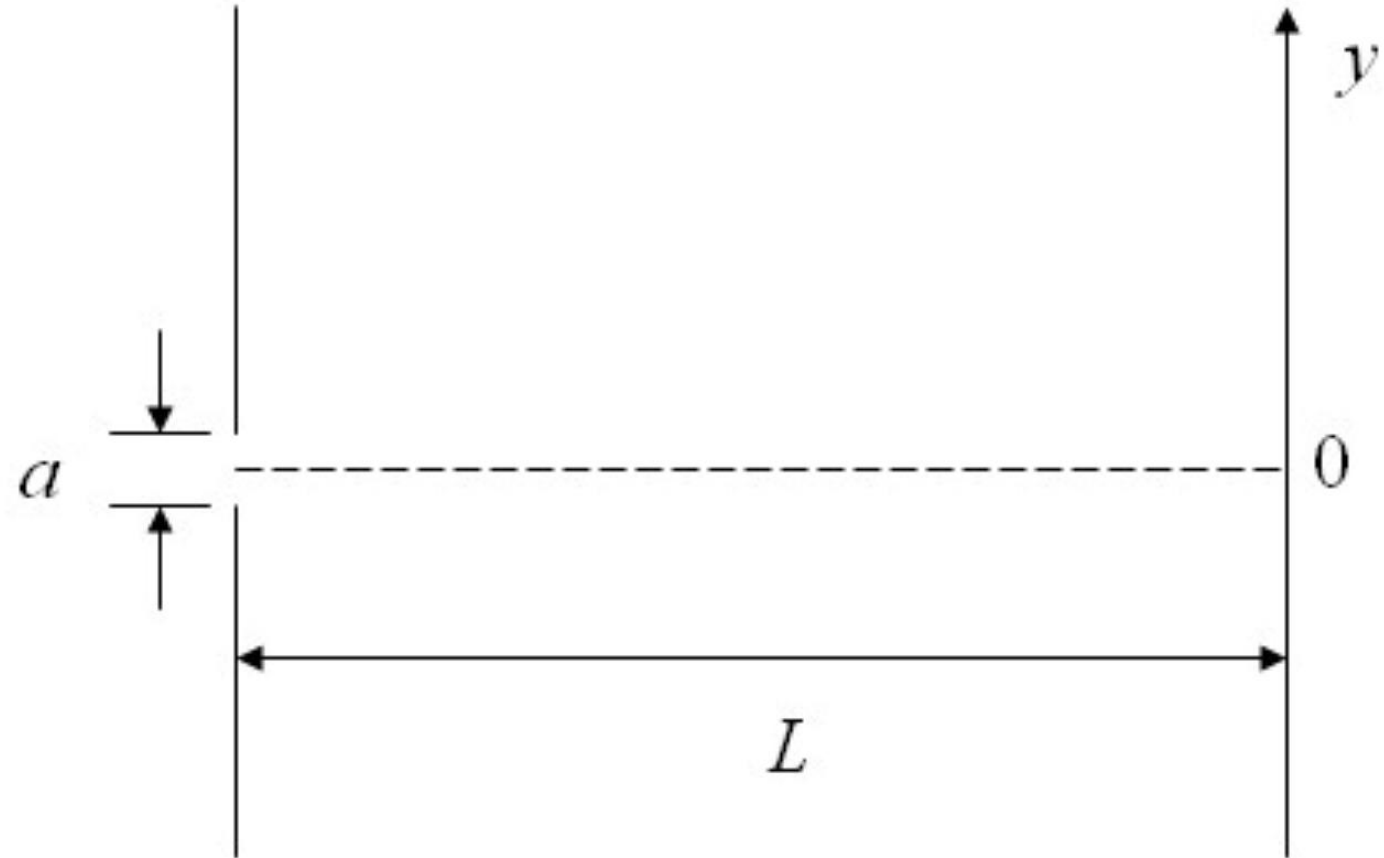
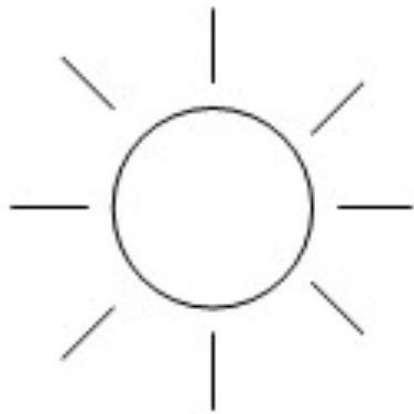
Fernfeld



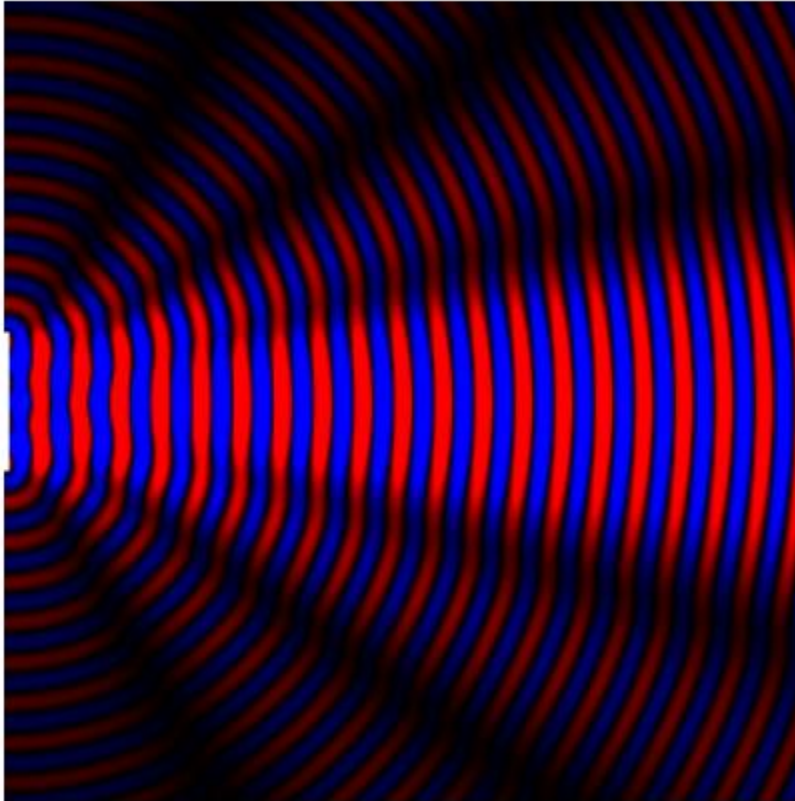
Konstruktive Interferenz: $|\vec{r}_2| - |\vec{r}_1| = n\lambda \approx d \sin \theta \approx \frac{yd}{L}$

Destruktive Interferenz: $|\vec{r}_2| - |\vec{r}_1| = (n + \frac{1}{2})\lambda \approx d \sin \theta \approx \frac{yd}{L}$

Einfachspalt

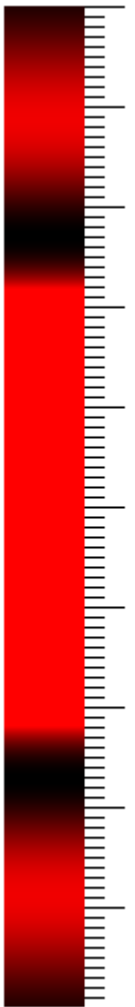


Einfachspalt

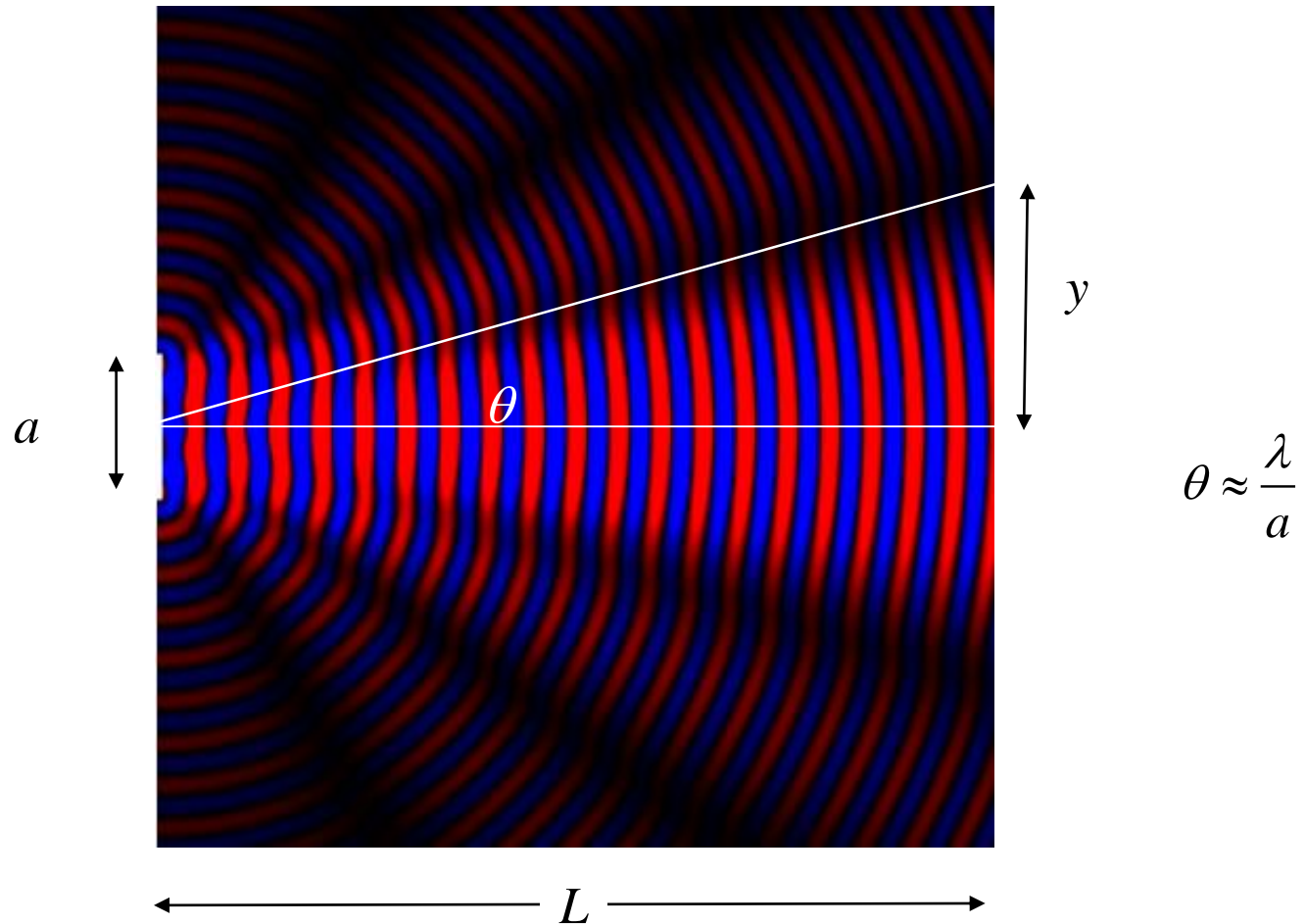


$$N = 40$$
$$\lambda = 0.3 \text{ [cm]}$$
$$a = 1 \text{ [cm]}$$
$$T = 0.5 \text{ [s]}$$

plot bei $t = 0$ [s].
t - T/10 t + T/10

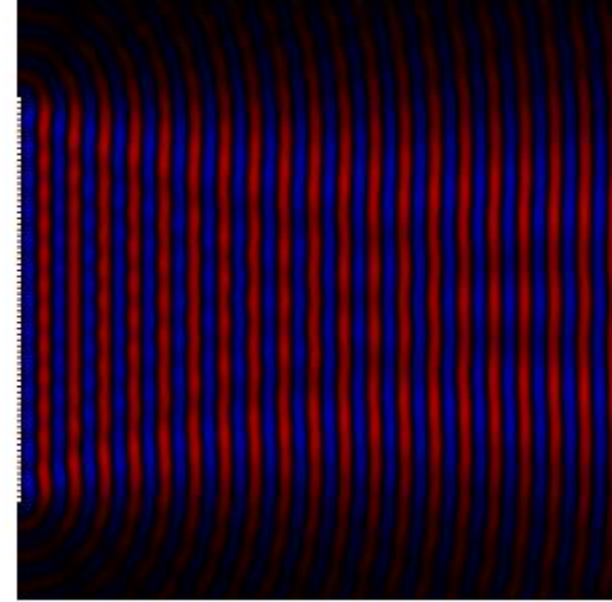
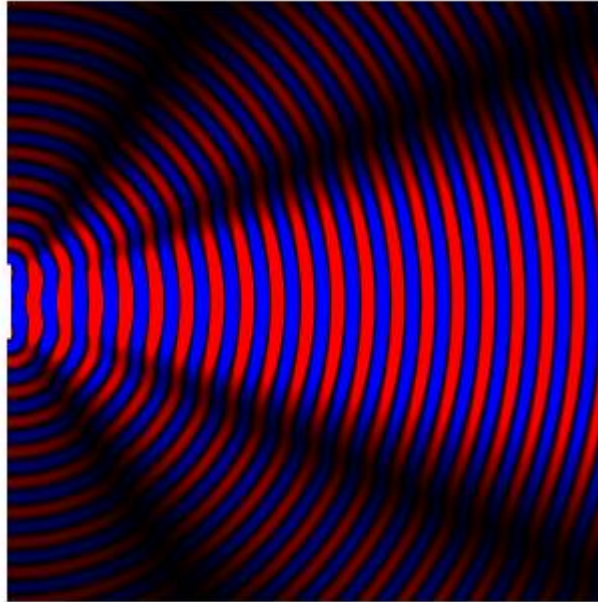
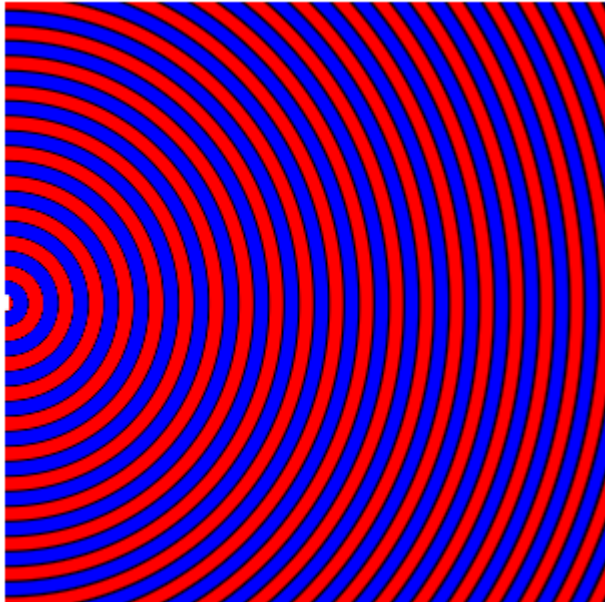


Einfachspalt Beugung

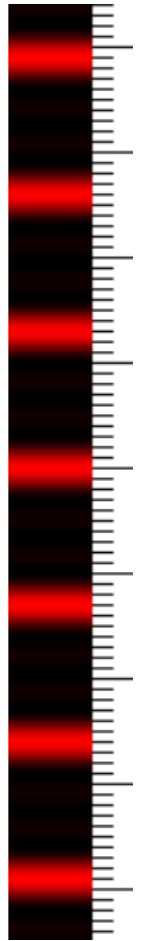
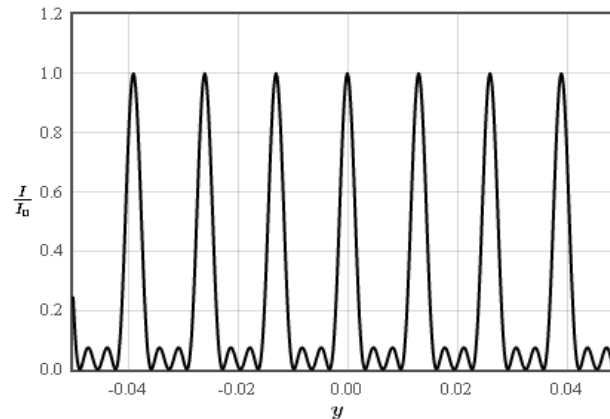
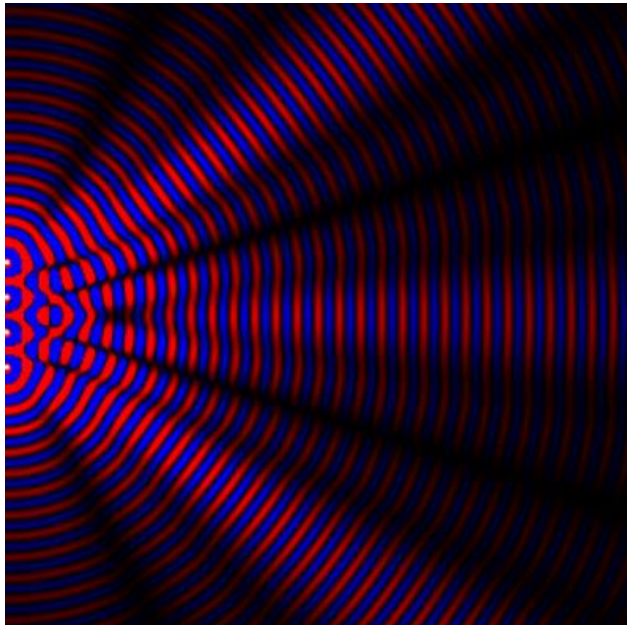
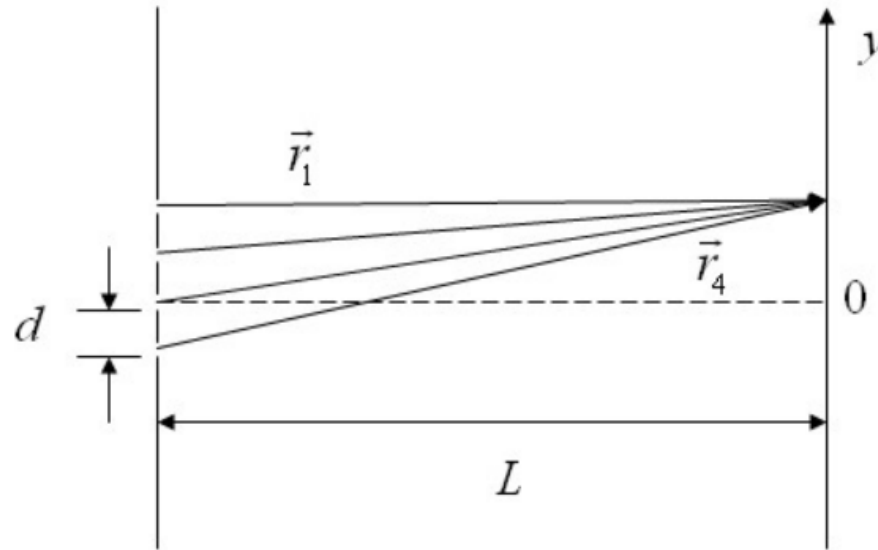
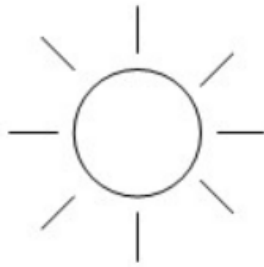


Destruktive Interferenz: $|\vec{r}_2| - |\vec{r}_1| = \frac{\lambda}{2} \approx \frac{a}{2} \sin \theta \approx \frac{ya}{2L}$
(Fernfeld)

Beugung



Interferenz am N -fach Spalt



Konstruktive Interferenz: $|\vec{r}_2| - |\vec{r}_1| = n\lambda \approx d \sin \theta \approx \frac{yd}{L}$
(Fernfeld)